ME တက်တဲ့ကျောင်းသားတွေဟာ Textbook/ Problem Solution စတာတွေကိုဆရာအကူအညီမပါဘဲမိမိ ဘာသာဖတ်ပြီးနားလည်တာကိုအကျယ်တဝင့်ရေးသားဖေါ် ပြနိုင်လိုရင်းကိုအတိုချုပ်ရေးသားပြီး အဓိက အချက်တွေ၊ဖေါ် မြူလာတွေ၊လက်တွေ့ အသုံးချချက်တွေကိုကောက်နှုတ်နိုင်စွမ်းရှိရမယ်။ အဲဒီစွမ်းရည်ရအောင်လေ့ကျင့်နည်းတွေကိုအခုထဲ့သွင်းထားတဲ့လင့်တွေမှာပြထားတယ်။ Handbook of Engineering Calculations ဟာတကယ်လက်တွေ့ အင်ဂျင်နီယာဒီဇိုင်းပုစ္ဆာတွေ ကိုဘယ်လိုဖြေရှင်းတွက်ချက်ရမယ်ဆိုတာနည်းလမ်းပေါင်းစုံပြထားတယ်။ အဲဒါကိုဘယ်လိုအကျယ်တဝင့်ရှင်းလင်းမယ်၊လိုအပ်တဲ့ Data တွေ၊ဇယားတွေ၊ဝေါဟာရတွေကို အင်တာနက်ကဘယ်လိုရာရမယ်ဆိုတာ Mechanical/ Civil/ Electricalနမူနာ၃ခုပြထားတယ်။

www.iqytechnicalcollege.com/MEquality.pdf

IQY ရဲ့ Master Diploma Course မှာစာမေးပွဲမထားဘဲကျောင်းသားတွေအဲဒီလိုအဖြေရှာ၊နားလည် ချက်ကိုတင်ပြချက်တွေကိုသုံးသပ်အကဲဖြတ်တဲ့နည်းကိုသုံးပါတယ်။ အဲဒါဟာနက်နက်နဲနဲစဉ်းစားတွေးခေါ်ခြင်းအလေ့အကျင့်ကိုအင်ဂျင်နီယာပညာရေးမှာသုံးထားတဲ့အလေ့အ ကျင့်ပြုနည်းနမူနာတစ်ခုဖြစ်ပါတယ်။

Critical Thinking Video

https://youtu.be/Cekuc04E2xM

The students who are attending Master of Engineering degree program need to possess the skills that they can read and understand the textbooks and problem solutions without assistance provided by teacher and write what understood, express the main points, highlight the key formulae and indicate the way to be utilize in the practice.

Handbook of Engineering Calculations contains the summarized advice on how to solve many practical design problems.

The students need to perform the expanded calculations based on the suggested guideline solution advice by collecting the required data, facts, tables extracts etc from internet.

The examples on how to perform such task is provided in three worked examples in mechanical, civil and electrical engineering.

Master Diploma course of IQY Technical College does not have the examination instead the evaluation and assessment on how the students find the solution and their understanding

It is one of the examples on applications of critical thinking exercises in engineering education.

BAE 701 Engineering Fundamental

Civil

www.mongroupsydney1.com/1.pdf

select one specialisation

Section 1-Civil Engineering (PDFFile Page 7)
Section 6- Water & Waste Water Engineering (PDF File Page 1041)
Section 7-Environmental Engineering (PDF File Page 1078) 2

For every topic, you need to write the short note on what you understand, formula, summary, outlines and at least 2 problems solution (Please note, each problem is solved in short form, you need to clearly reproduce them by step by step)

Section 1-Civil Engineering (PDFFile Page 7)

STRUCTURAL STEEL DESIGN

Steel Beams and Plate Girders

Notes

The notational system used conforms with that given, and it is augmented to include the following:

```
Aw = area of flange, in2 (cm2);

Aw = area of web, in2 (cm2);

bf = width of flange, in (mm);

d = depth of section, in (mm);

dw = depth of web, in (mm);

tf = thickness of flange, in (mm);

tw = thicknessof web, in (mm);

L' = unbraced length of compression flange, in (mm);

fy = yield-point stress,

lb/in2 (kPa).
```

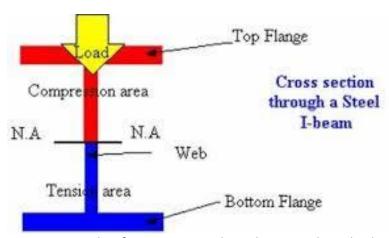
Flange?---- Internet search

A flange is an external or internal ridge, or rim, for strength, as the flange of an iron beam such as an I-beam or a T-beam; or for attachment to another object, as the flange on the end of a pipe



Web? Internet search

Images for area of web i beam



It is an example of every new technical term is described and appropriate internet research is performed so that you can understand the exact meaning.

Problem

A beam on a simple span of 30 ft (9.2 m) carries a uniform superimposed load of 1650 lb/lin ft(24,079.9 N/m). The compression flange is laterally supported along its entire length. Select the most economic section.

Calculation Procedure in the book

1. Compute the maximum bending moment and the required section modulus. Assume that the beam weighs 50 lb/lin ft (729.7 N/m) and satisfies the requirements of a compact section as set forth in the *Specification*.

Referring to the *Specification* shows that the allowable bending stress is 24,000 lb/in2 (165,480.0 kPa). Then S = M/f = 2,295,000/24,000 = 95.6 in 3 (1566.88 cm 3).

2. Select the most economic section. Refer to the AISC *Manual*, and select the most economic section. Use W18 \times 55 = 98.2 in3 (1609.50 cm3); section compact. The disparity between the assumed and actual beam weight is negligible.

A second method for making this selection is shown below.

- **3.** Calculate the total load on the member. Thus, the total load = W = 30(1700) = 51,000 lb (226,848.0 N).
- **4. Select the most economic section.** Refer to the tables of allowable uniform loads in the *Manual*, and select the most economic section. Thus, use W18 \times 55; Wallow = 52,000 lb (231,296.0 N). The capacity of the beam is therefore slightly greater than required.

Detailed explanation to be done (Worked Example)

Superimposed load-/ Maximum bending moment/ Section modulus -- Internet search



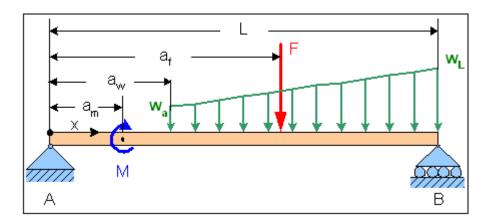
Types of Loads

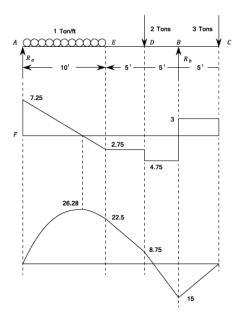
4. Superimposed Load

- This term is used for all external loads, leaving the self weight, acting on the member to be designed.
- This includes live load, wind load, earthquake load, etc. Part of dead load may also act as imposed load.

5. Service Load

- The maximum intensity of load expected during the life of the structure depending upon a certain probability of occurrence is called service load.
- No additional factor of safety or overload factor is included in the service loads.





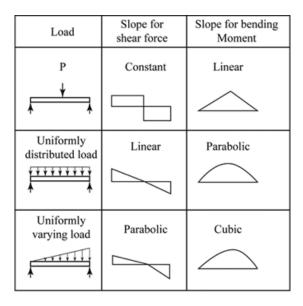


Figure-1 Slopes for various types of loads

Section modulus is a geometric property for a given cross-**section** used in the design of beams or flexural members. Other geometric properties used in design include area for tension and shear, radius of gyration for compression, and moment of inertia and polar moment of inertia for stiffness.

Calculating the section modulus

To calculate the section modulus, the following formula applies:

$$Z = \frac{1}{y}$$
 where $I =$ moment of inertia, $y =$ distance from centroid to top or bottom edge of the rectangle

For symmetrical sections the value of **Z** is the same above or below the centroid.

For asymmetrical sections, two values are found: **Z** max and **Z** min.

To calculate the value of **Z** for a simple symmetrical shape such as a rectangle:

$$Z_{XX} = \frac{Ixx}{y}$$
 where $I_{XX} = \frac{bd^3}{12} \text{ mm}^4$

$$\frac{1/2}{2} \text{ depth or } \frac{d}{2} \text{ mm}$$
 and y =

$$Z = \frac{bd^2}{6} \text{ mm}^3$$

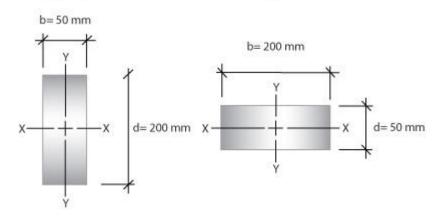
This gives the formula for **Z** as:

Note: The standard form of writing the value of $\bf Z$ is to write it as a number x 10^3 mm³, eg a value of 2,086 is written as 2.086×10^3 .

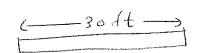
Calculating Z



Beam B



WORKED EXAMPLE FOR GIVEN PROBLEM AND SHORT SOLUTION PROCESS



Superimposed Load = 1650 lb/Lim ft

maximum bending moment. $M = \frac{Wl^2}{8} ft-ls$ $M = \frac{Wl^2}{8} x_{12} \text{ in-ls}$

 $\frac{1}{1} = \frac{1700 \times 30^{2} \times 12}{8} = 2245000 \text{ in lb}$ $\frac{1}{1} = 0.113 \text{ N-m} \quad \text{(Intermed Fearch)}$ $\frac{1}{1} = 2245000 \text{ in lb} \Rightarrow 2245000 \times 0.113$

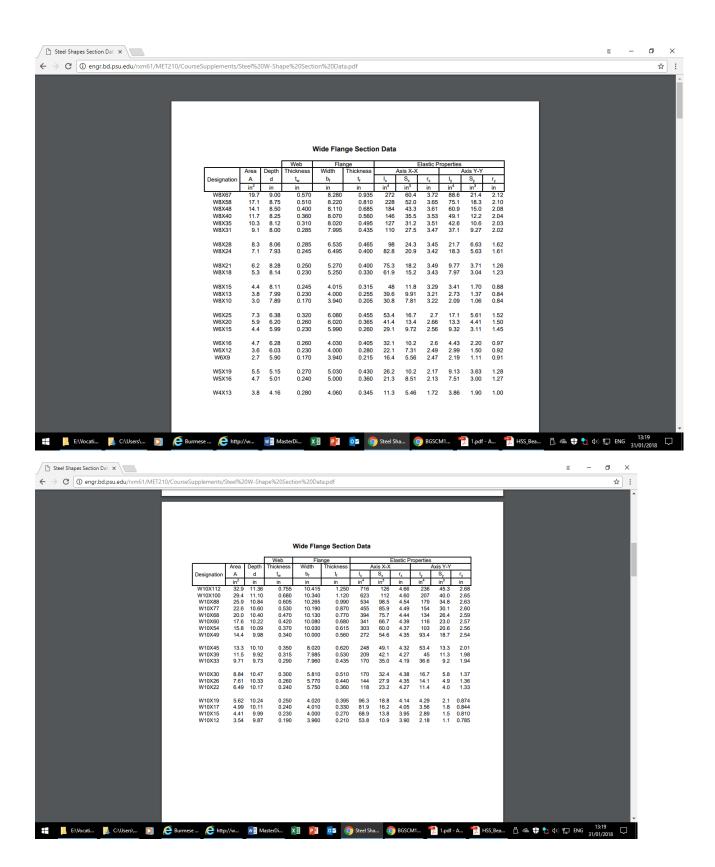
= 259289 N-m

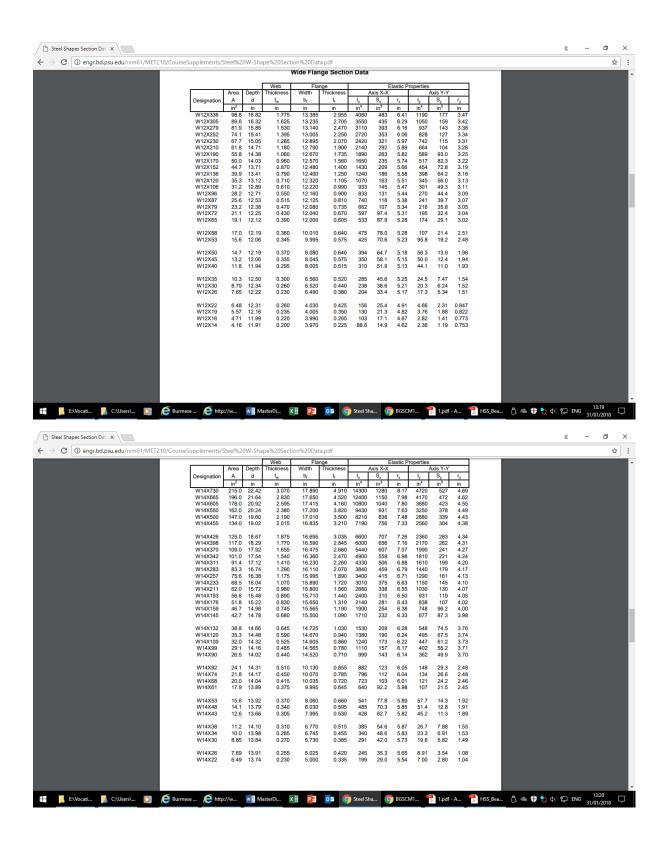
Allowable bending stress (as per American Steel standard specification)

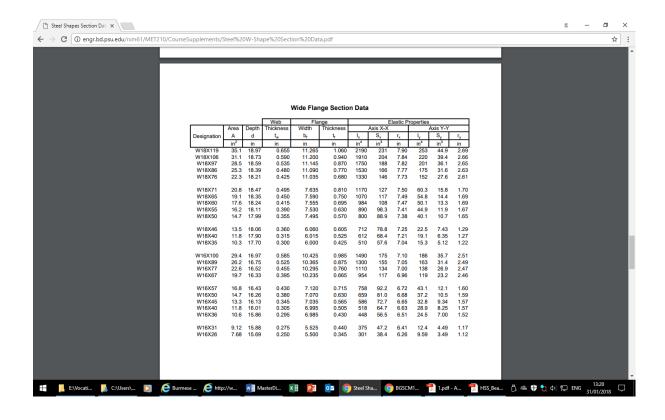
Index (S) = Bending moment
maximum Bending Stress

 $S = \frac{m}{f} = \frac{2295000}{24000} = 95.6 \text{ im}^3$ (or) (1566.88)

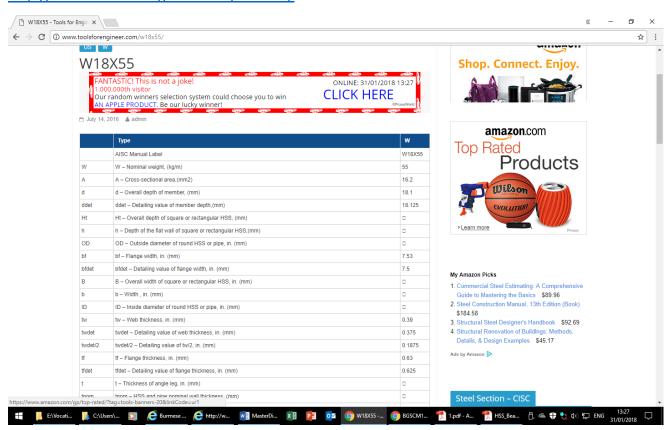
http://engr.bd.psu.edu/rxm61/MET210/CourseSupplements/Steel%20W-Shape%20Section%20Data.pdf







http://www.toolsforengineer.com/w18x55/



							Static Parameters					
Designation			Dir	Moment of Inertia		Elastic Section Modulus						
Imperial (in x lb/ft)	Depth h (in)	Width w (in)	Web Thickness t _w (in)	Flange Thickness t _f (in)	Sectional Area (in²)	Weight (Ib/ft)	l _x (in⁴)	l _y (in⁴)	W _x (in³)	W _y (in³)		
W 27 x 178	27.8	14.09	0.725	1.190	52.3	178	6990	555	502	78.8		
W 27 x 161	27.6	14.02	0.660	1.080	47.4	161	6280	497	455	70.9		
W 27 x 146	27.4	14	0.605	0.975	42.9	146	5630	443	411	63.5		
W 27 x 114	27.3	10.07	0.570	0.930	33.5	114	4090	159	299	31.5		
W 27 x 102	27.1	10.02	0.515	0.830	30.0	102	3620	139	267	27.8		
W 27 x 94	26.9	10	0.490	0.745	27.7	94	3270	124	243	24.8		
W 27 x 84	26.7	9.96	0.460	0.640	24.8	84	2850	106	213	21.2		
W 24 x 162	25	13	0.705	1.220	47.7	162	5170	443	414	68.4		
W 24 x 146	24.7	12.9	0.650	1.090	43.0	146	4580	391	371	60.5		
W 24 x 131	24.5	12.9	0.605	0.960	38.5	131	4020	340	329	53.0		
W 24 x 117	24.3	12.8	0.55	0.850	34.4	117	3540	297	291	46.5		
W 24 x 104	24.1	12.75	0.500	0.750	30.6	104	3100	259	258	40.7		
W 24 x 94	24.1	9.07	0.515	0.875	27.7	94	2700	109	222	24.0		
W 24 x 84	24.1	9.02	0.470	0.770	24.7	84	2370	94.4	196	20.9		

						Static F	Paramete	ers		
Designation			Din	nensions			Moment of Inertia		Elastic Section Modulus	
Imperial (in x lb/ft)	Depth h (in)	Width w (in)	Web Thickness t _w (in)	Flange Thickness t _f (in)	Sectional Area (in²)	Weight (Ib/ft)	l _x (in⁴)	l _y (in⁴)	W _x (in³)	W _y (in³)
W 24 x 76	23.9	9	0.440	0.680	22.4	76	2100	82.5	176	18.4
W 24 x 68	23.7	8.97	0.415	0.585	20.1	68	1830	70.4	154	15.7
W 24 x 62	23.7	7.04	0.430	0.590	18.2	62	1550	34.5	131	9.8
W 24 x 55	23.6	7.01	0.395	0.505	16.2	55	1350	29.1	114	8.3
W 21 x 147	22.1	12.51	0.720	1.150	43.2	147	3630	376	329	60.1
W 21 x 132	21.8	12.44	0.650	1.035	38.8	132	3220	333	295	53.5
W 21 x 122	21.7	12.39	0.600	0.960	35.9	122	2960	305	273	49.2
W 21 x 111	21.5	12.34	0.550	0.875	32.7	111	2670	274	249	44.5
W 21 x 101	21.4	12.29	0.500	0.800	29.8	101	2420	248	227	40.3
W 21 x 93	21.6	8.42	0.580	0.930	27.3	93	2070	92.9	192	22.1
W 21 x 83	21.4	8.36	0.515	0.835	24.3	83	1830	81.4	171	19.5
W 21 x 73	21.2	8.3	0.455	0.740	21.5	73	1600	70.6	151	17.0
W 21 x 68	21.1	8.27	0.430	0.685	20.0	68	1480	64.7	140	15.7
W 21 x 62	21	8.24	0.400	0.615	18.3	62	1330	57.5	127	13.9
W 21 x 57	21.1	6.56	0.405	0.650	16.7	57	1170	30.6	111	9.4
W 21 x 50	20.8	6.53	0.380	0.535	14.7	50	984	24.9	94.5	7.6
W 21 x 44	20.7	6.5	0.350	0.450	13.0	44	843	20.7	81.6	6.4

								Static F	Paramete	ers
Designation			Dir	nensions			Moment of Inertia		Elastic Section Modulus	
Imperial (in x lb/ft)	Depth h (in)	Width w (in)	Web Thickness t _w (in)	Flange Thickness t _f (in)	Sectional Area (in²)	Weight (Ib/ft)	l _x (in⁴)	ly (in⁴)	W _x (in³)	W _y (in³)
W 18 x 119	19	11.27	0.655	1.060	35.1	119	2190	253	231	44.9
W 18 x 106	18.7	11.2	0.590	0.940	31.1	106	1910	220	204	39.4
W 18 x 97	18.6	11.15	0.535	0.870	28.5	97	1750	201	188	36.1
W 18 x 86	18.4	11.09	0.480	0.770	25.3	86	1530	175	166	31.6
W 18 x 76	18.2	11.04	0.425	0.680	22.3	76	1330	152	146	27.6
W 18 x 71	18.5	7.64	0.495	0.810	20.8	71	1170	60.3	127	15.8
W 18 x 65	18.4	7.59	0.450	0.750	19.1	65	1070	54.8	117	14.4
W 18 x 60	18.2	7.56	0.415	0.695	17.6	60	984	50.1	108	13.3
W 18 x 55	18.1	7.53	0.390	0.630	16.2	55	890	44.9	98.3	11.9
W 18 x 50	18	7.5	0.355	0.570	14.7	50	800	40.1	88.9	10.7
W 18 x 46	18.1	6.06	0.360	0.605	13.5	46	712	22.5	78.8	7.4
W 18 x 40	17.9	6.02	0.315	0.525	11.8	40	612	19.1	68.4	6.4
W 18 x 35	17.7	6	0.300	0.425	10.3	35	510	15.3	57.6	5.1
W 16 x 100	16.97	10.425	0.585	0.985	29.4	100	1490	186	175	35.7
W 16 x 89	16.75	10.365	0.525	0.875	26.2	89	1300	163	155	31.4
W 16 x 77	16.52	10.295	0.455	0.760	22.6	77	1100	138	134	26.9

					Static F	Paramete	ers			
Designation			Dir	nensions			Moment of Inertia		Elastic Mod	Section ulus
Imperial (in x lb/ft)	Depth h (in)	Width w (in)	Web Thickness t _w (in)	Flange Thickness t _f (in)	Sectional Area (in²)	Weight (Ib/ft)	l _x (in⁴)	l _y (in⁴)	W _x (in³)	W _y (in³)
W 16 x 67	16.33	10.235	0.395	0.665	19.7	67	954	119	117	23.2
W 16 x 57	16.43	7.120	0.430	0.715	16.8	57	758	43.1	92.2	12.1
W 16 x 50	16.26	7.070	0.380	0.630	14.7	50	659	37.2	81	10.5
W 16 x 45	16.13	7.035	0.345	0.565	13.3	45	586	32.8	72.7	9.3
W 16 x 40	16.01	6.995	0.305	0.505	11.8	40	518	28.9	64.7	8.3
W 16 x 36	15.86	6.985	0.295	0.430	10.6	36	448	24.5	56.5	7
W 16 x 31	15.88	5.525	0.275	0.440	9.12	31	375	12.4	47.2	4.5
W 16 x 26	15.69	5.5	0.250	0.345	7.68	26	301	9.6	38.4	3.5
W 14 x 132	14.66	14.725	0.645	1.030	38.8	132	1530	548	209	74.5
W 14 x 120	14.48	14.670	0.590	0.940	35.3	120	1380	495	190	67.5
W 14 x 109	14.32	14.605	0.525	0.860	32	109	1240	447	173	61.2
W 14 x 99	14.16	14.565	0.485	0.780	29.1	99	1110	402	157	55.2
W 14 x 90	14.02	14.520	0.440	0.710	26.5	90	999	362	143	49.9
W 14 x 82	14.31	10.130	0.510	0.855	24.1	82	882	148	123	29.3
W 14 x 74	14.17	10.070	0.450	0.785	21.8	74	796	134	112	26.6
W 14 x 68	14.04	10.035	0.415	0.720	20.0	68	723	121	103	24.2
W 14 x 61	13.89	9.995	0.375	0.645	17.9	61	640	107	92.2	21.5

				Static Parameters						
Designation			Dir	nensions			Moment of Inertia		Elastic Mod	Section ulus
Imperial (in x lb/ft)	Depth h (in)	Width w (in)	Web Thickness t _w (in)	Flange Thickness t _f (in)	Sectional Area (in²)	Weight (Ib/ft)	l _x (in⁴)	l _y (in⁴)	W _x (in³)	W _y (in³)
W 14 x 53	13.92	8.060	0.370	0.660	15.6	53	541	57.7	77.8	14.3
W 14 x 48	13.79	8.030	0.340	0.595	14.1	48	485	51.4	70.3	12.8
W 14 x 43	13.66	7.995	0.305	0.530	12.6	43	428	45.2	62.7	11.3
W 14 x 38	14.10	6.770	0.310	0.515	11.2	38	385	26.7	54.6	7.9
W 14 x 34	13.98	6.745	0.285	0.455	10.0	34	340	23.3	48.6	6.9
W 14 x 30	13.84	6.730	0.270	0.385	8.85	30	291	19.6	42.0	5.8
W 14 x 26	13.91	5.025	0.255	0.420	7.69	26	245	8.9	35.3	3.5
W 14 x 22	13.74	5	0.230	0.335	6.49	22	199	7	29.0	2.8
W 12 x 136	13.41	12.4	0.79	1.250	39.9	136	1240	398	186	64.2
W 12 x 120	13.12	12.32	0.71	1.105	35.3	120	1070	345	163	56.0
W 12 x 106	12.89	12.22	0.61	0.990	31.2	106	933	301	145	49.3
W 12 x 96	12.71	12.16	0.55	0.900	28.2	96	833	270	131	44.4
W 12 x 87	12.53	12.125	0.515	0.810	25.6	87	740	241	118	39.7
W 12 x 79	12.38	12.08	0.47	0.735	23.2	79	662	216	107	35.8
W 12 x 72	12.25	12.04	0.43	0.670	21.1	72	597	195	97.4	32.4
W 12 x 65	12.12	12	0.39	0.605	19.1	65	533	174	87.9	29.1
W 12 x 58	12.19	10.01	0.36	0.640	17.0	58	475	107	78	21.4

				Static Parameters						
Designation			Dir	nensions			Moment of Inertia		Elastic Mod	
Imperial (in x lb/ft)	Depth h (in)	Width w (in)	Web Thickness t _w (in)	Flange Thickness t _f (in)	Sectional Area (in²)	Weight (Ib/ft)	l _x (in⁴)	l _y (in⁴)	W _x (in³)	W _y (in³)
W 12 x 53	12.06	9.995	0.345	0.575	15.6	53	425	95.8	70.6	19.2
W 12 x 50	12.19	8.08	0.37	0.640	14.7	50	394	56.3	64.7	13.9
W 12 x 45	12.06	8.045	0.335	0.575	13.2	45	350	50.0	58.1	12.4
W 12 x 40	11.94	8.005	0.295	0.515	11.8	40	310	44.1	51.9	11.0
W 12 x 35	12.50	6.56	0.3	0.520	10.3	35	285	24.5	45.6	7.5
W 12 x 30	12.34	6.52	0.26	0.440	8.8	30	238	20.3	38.6	6.2
W 12 x 26	12.22	6.490	0.23	0.380	7.7	26	204	17.3	33.4	5.3
W 12 x 22	12.31	4.03	0.26	0.425	6.5	22	156	4.7	25.4	2.3
W 12 x 19	12.16	4.005	0.235	0.350	5.6	19	130	3.8	21.3	1.9
W 12 x 16	11.99	3.990	0.22	0.265	4.7	16	103	2.8	17.1	1.4
W 12 x 14	11.91	3.970	0.2	0.225	4.2	14	88.6	2.4	14.9	1.2
W 10 x 112	11.36	10.415	0.755	1.250	32.9	112	716	236	126	45.3
W 10 x 100	11.1	10.340	0.680	1.1120	29.4	100	623	207	112	40.0
W 10 x 88	10.84	10.265	0.605	0.990	25.9	88	534	179	98.5	34.8
W 10 x 77	10.60	10.190	0.530	0.870	22.6	77	455	154	85.9	30.1
W 10 x 68	10.40	10.130	0.470	0.770	20.0	68	394	134	75.7	26.4
W 10 x 60	10.22	10.080	0.420	0.680	17.6	60	341	116	66.7	23.0

					Static F	Paramete	ers			
Designation			Din	nensions			Moment of Inertia		Elastic Mod	Section ulus
Imperial (in x lb/ft)	Depth h (in)	Width w (in)	Web Thickness t _w (in)	Flange Thickness t _f (in)	Sectional Area (in²)	Weight (Ib/ft)	l _x (in⁴)	l _y (in⁴)	W _x (in³)	W _y (in³)
W 10 x 54	10.09	10.030	0.370	0.615	15.8	54	303	103	60.0	20.6
W 10 x 49	9.98	10	0.340	0.560	14.4	49	272	93.4	54.6	18.7
W 10 x 45	10.10	8.020	0.350	0.620	13.3	45	248	53.4	49.1	13.3
W 10 x 39	9.92	7.985	0.315	0.530	11.5	39	209	45.0	42.1	11.3
W 10 x 33	9.73	7.960	0.290	0.435	9.71	33	170	36.6	35.0	9.2
W 10 x 30	10.47	5.81	0.3	0.510	8.84	30	170	16.7	32.4	5.8
W 10 x 26	10.33	5.770	0.26	0.440	7.6	26	144	14.1	27.9	4.9
W 10 x 22	10.17	5.750	0.240	0.360	6.5	22	118	11.4	23.2	4
W 10 x 19	10.24	4.020	0.250	0.395	5.6	19	96.3	4.3	18.8	2.1
W 10 x 17	10.11	4.010	0.240	0.330	5	17	81.9	3.6	16.2	1.8
W 10 x 15	9.99	4	0.230	0.270	4.4	15	68.9	2.9	13.8	1.5
W 10 x 12	9.87	3.960	0.190	0.210	3.5	12	53.8	2.2	10.9	1.1
W 8 x 67	9.00	8.280	0.570	0.935	19.7	67	272	88.6	60.4	21.4
W 8 x 58	8.75	8.220	0.510	0.810	17.1	58	228	75.1	52.0	18.3
W 8 x 48	8.5	8.110	0.400	0.685	14.1	48	184	60.9	43.3	15.0
W 8 x 40	8.25	8.070	0.360	0.560	11.7	40	146	49.1	35.5	12.2
W 8 x 35	8.12	8.020	0.310	0.495	10.3	35	127	42.6	31.2	10.6

				Static Parameters						
Designation			Dir	nensions				ent of rtia	Elastic Mod	Section ulus
Imperial (in x lb/ft)	Depth h (in)	Width w (in)	Web Thickness t _w (in)	Flange Thickness t _f (in)	Sectional Area (in²)	Weight (Ib/ft)	l _x (in⁴)	l _y (in⁴)	W _x (in³)	W _y (in³)
W 8 x 31	8.00	7.995	0.285	0.435	9.1	31	110	37.1	27.5	9.3
W 8 x 28	8.06	6.535	0.285	0.465	8.3	28	98.0	21.7	24.3	6.6
W 8 x 24	7.93	6.495	0.245	0.400	7.1	24	82.8	18.3	20.9	5.6
W 8 x 21	8.28	5.270	0.250	0.400	6.2	21	75.3	9.8	18.2	3.7
W 8 x 18	8.14	5.250	0.230	0.330	5.3	18	61.9	8	15.2	3
W 8 x 15	8.11	4.015	0.245	0.315	4.4	15	48.0	3.4	11.8	1.7
W 8 x 13	7.99	4	0.230	0.255	3.8	13	39.6	2.7	9.9	1.4
W 8 x 10	7.89	3.940	0.170	0.205	2.9	10	30.3	2.1	7.8	1.1
W 6 x 25	6.38	6.080	0.320	0.455	7.3	25	53.4	17.1	16.7	5.6
W 6 x 20	6.20	6.020	0.260	0.365	5.9	20	41.4	13.3	13.4	4.4
W 6 x 16	6.28	4.030	0.260	0.405	4.7	16	32.1	4.4	10.2	2.2
W 6 x 15	5.99	5.990	0.230	0.260	4.4	15	29.1	9.3	9.7	3.1
W 6 x 12	6.03	4	0.230	0.280	3.6	12	22.1	3	7.3	1.5
W 6 x 9	5.90	3.940	0.170	0.215	2.7	9	16.4	2.2	5.6	1.1
W 5 x 19	5.15	5.030	0.270	0.430	5.5	19	26.2	9.1	10.2	3.6
W 5 x 16	5.01	5	0.240	0.360	4.7	16	21.3	7.5	8.5	3

								Static Parameters				
Designation		Dimensions							Elastic Section Modulus			
Imperial (in x lb/ft)	Depth h (in)	h w mickness mickness Area weight						l _y (in⁴)	W _x (in³)	W _y (in³)		
W 4 x 13	4.16	4.060	0.280	0.345	3.8	13	11.3	3.9	5.5	1.9		

Bated on 95.6 im³ (1566.82 cm³)

Select the most economic Section by referring

ALSC Table Steel Shape (section data.

W12 X55 has 98.3 im³. Use it.

In the above worked example, based on solution outline

- Internet research was performed to exactly know the meaning of technical terms/ formula
- Internet research was performed to exactly know the details of materials usage
- Internet research was performed to exactly know the details technical table.

Although you may not have the AISC Manual, by doing appropriate search, you can have the necessary technical data.

- By using such data, the details and simplified calculation/ solution was made.
- The statement in the original outline solution was more detailly explained.

This is my worked example for **STRUCTURAL STEEL DESIGN**

BAE701 Engineering Fundamentals

Mechanical

www.mongroupsydney1.com/1.pdf

Then study Section 3-Mechanical Engineering (PDF File Page 307)

For every topic, you need to write the short note on what you understand, formula, summary, outlines and at least 2 problems solution (Please note, each problem is solved in short form, you need to clearly reproduce them by step by step)

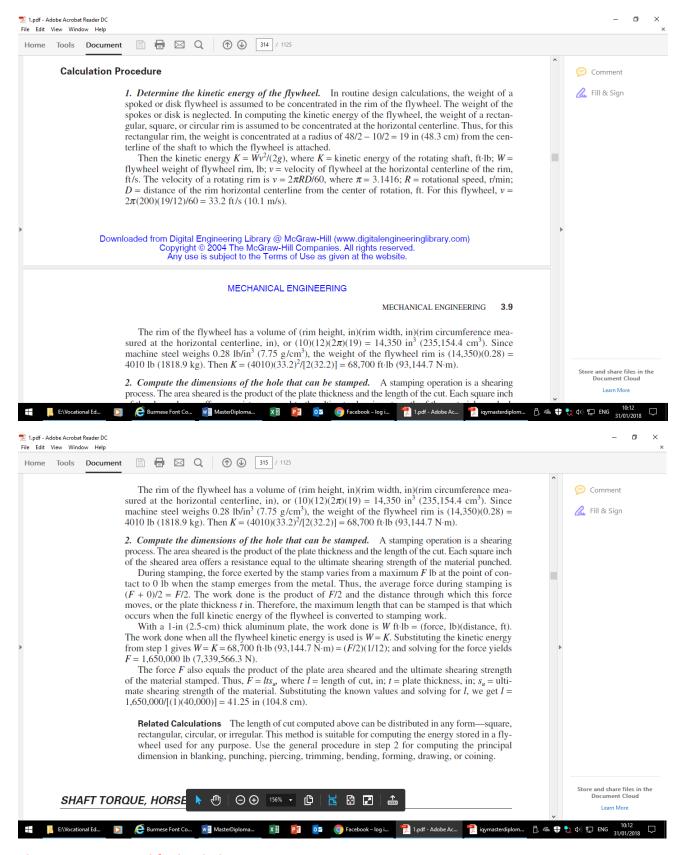
MECHANICAL DESIGN AND ANALYSIS 3.8

Page 314 Energy Stored in Rotating Flywheel

The Questions and suggested solution in the book

A 48-in (121.9-cm) diameter spoked steel flywheel having a 12-in wide \times 10-in (30.5-cm \times 25.4-cm) deep rim rotates at 200 r/min. How long a cut can be stamped in a 1-in (2.5-cm) thick aluminium plate if the stamping energy is obtained from this flywheel? The ultimate shearing strength of the aluminum is 40,000 lb/in2 (275,789.9 kPa).

Procedure written in the book



That process is simplified as below

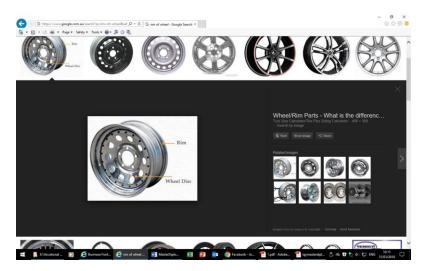
Step 1

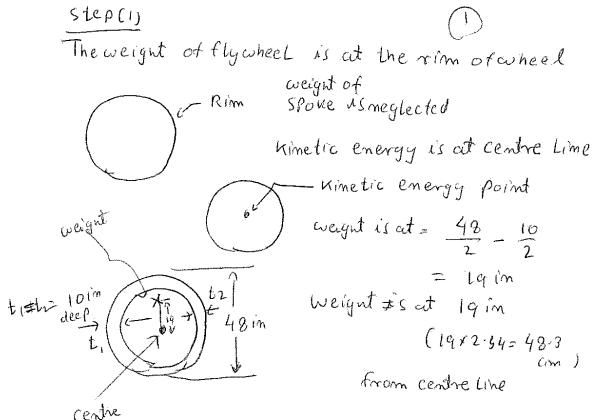
1. Determine the kinetic energy of the flywheel. In routine design calculations, the weight of a spoked or disk flywheel is assumed to be concentrated in the rim of the flywheel. The weight of the

spokes or disk is neglected. In computing the kinetic energy of the flywheel, the weight of a rectangular, square, or circular rim is assumed to be concentrated at the horizontal centerline. Thus, for this

rectangular rim, the weight is concentrated at a radius of 48/2 10/2 = 19 in (48.3 cm) from the centreline of the shaft to which the flywheel is attached.

The weight of flywheel is at the rim of wheel (rim of wheel? Internet search –





Step 2

Then the kinetic energy $K = W\nu 2/(2g)$, where K = kinetic energy of the rotating shaft, ft \oplus lb; W =flywheel weight of flywheel rim, lb; $\nu =$ velocity of flywheel at the horizontal centerline of the rim,

The velocity of a rotating rim is $v = 2\pi RD/60$, where $\pi = 3.1416$; R = rotational speed, r/min;

D =distance of the rim horizontal centerline from the center of rotation, ft. For this flywheel,

 $v = 2\pi(200) (19/12)/60 = 33.2 \text{ ft /s } (10.1 \text{ m/s}).$

$$\frac{\text{Step(2)}}{\text{Kimetic Energy}}$$
 $K = \frac{\text{Wv}^2}{2g}$

K= kimetic Energy of rotating shaft (ft-lb)

W = weight of fly wheel (lb)

v = velocity of flywheel at the horizontal

centre line of the rim (fl/s)

U = velocity of rotating rim

TI = 3.1416, R = Rotational Speed ofmin D= Distance of the rim horizontal centreline from centre of rotation (ft)

$$D = \frac{19 \text{ in}}{19 \text{ in}} = \frac{19 \text{ lin}}{60} = \frac{2 \times 3.1416 \times 200 \times \left(\frac{19 \text{ in}}{12}\right) \text{ ft}}{60}$$

$$= \frac{33.2 \text{ ft/sec}}{33.2 \text{ ft/sec}} = \frac{33.2}{3.3}$$

$$= \frac{10.1 \text{ m/c}}{60}$$

The rim of the flywheel has a volume of (rim height, in) (rim width, in) (rim circumference measured at the horizontal centerline, in), or $(10) (12) (2\pi) (19) = 14,350$ in (235,154.4 cm3).

Since machine steel weighs 0.28 lb/in3 (7.75 g/cm3), the weight of the flywheel rim is (14,350) (0.28) = 4010 lb (1818.9 kg). Then *K* = (4010) (33.2) 2/[2(32.2)] = 68,700 ft⊕lb (93,144.7 N⊕m).

٠. ٠ Step(3) volume of fly wheel Rim height = 10 Rim Weignt = 12 Rim circum ference - 271 (191 Recause it was given Crim height, in) (rim width, in) (rim corrunterence, 10 12 (27) (19) - 10x12x 2 Ti x19 = 14350 in3 (07) 1 in = 2-54(m $(4350 \times (2.54)^3 = 235154.4$ Steel weight 0,28 lb/in3 (Ur) 7-75,9m/cm? - weight of flywheel rim = 14350x0.28 1 Ky = 16 X0.456 = (4010X0-456) = 1212.9Kg

$$K = \frac{\omega u^{2}}{2g} = \frac{4010 \times 33.2^{2}}{2 \times 32.2} = 68790 \text{ ft-ly}$$

$$= \frac{1319.9 \times 10.2}{2 \times 9.8} = 93144.7$$
N-m

Step 4

2. Compute the dimensions of the hole that can be stamped.

A stamping operation is a shearing process. The area sheared is the product of the plate thickness and the length of the cut. Each square inch of the sheared area offers a resistance equal to the ultimate shearing strength of the material punched.

During stamping, the force exerted by the stamp varies from a maximum F lb at the point of contact to 0 lb when the stamp emerges from the metal. Thus, the average force during stamping is (F+0)/2 = F/2.

The work done is the product of F/2 and the distance through which this force moves, or the plate thickness t in. Therefore, the maximum length that can be stamped is that which occurs when the full kinetic energy of the flywheel is converted to stamping work.

With a 1-in (2.5-cm) thick aluminum plate, the work done is $W \text{ ft} \oplus \text{Ib} = (\text{force}, \text{Ib}) (\text{distance}, \text{ft}).$

The work done when all the flywheel kinetic energy is used is W = K. Substituting the kinetic energy

from step 1 gives W = K = 68,700 ft \oplus Ib (93,144.7 N \oplus m) = (F/2) (1/12); and solving for the force yields

F = 1,650,000 lb (7,339,566.3 N).

The force F also equals the product of the plate area sheared and the ultimate shearing strength

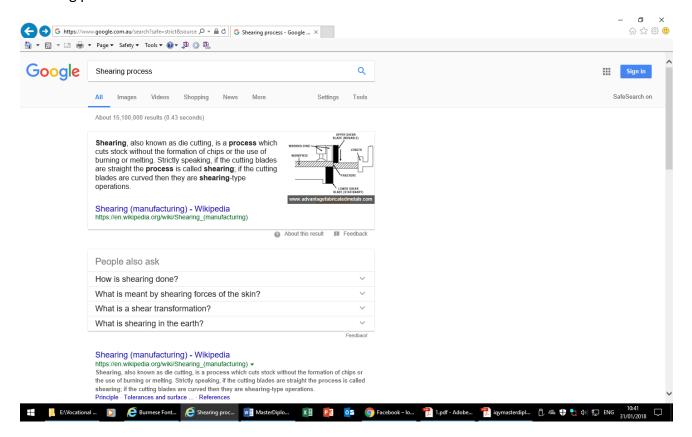
of the material stamped. Thus, F = ltsu, where l = length of cut, in; t = plate thickness, in; su = ultimate shearing strength of the material. Substituting the known values and solving for l,

we get l = 1,650,000/[(1)(40,000)] = 41.25 in (104.8 cm).

Note

A stamping operation is a shearing process. The area sheared is the product of the plate thickness and the length of the cut. Each square inch of the sheared area offers a resistance equal to the ultimate shearing strength of the material punched.

Shearing process—Internet search



Shearing Sheared Area.

Trickness of Plate

& Square in of Snewed Arrea = ultimate Sheeving Strength.

Average force during $=\frac{F+0}{2}=\frac{f}{2}$ Stamping

Because also at stampemerged from metal work done = Product of & and thickness & = $\frac{f}{2} \times t$

workdome when all flywheel kimetic Energy used W=K where K= 68700 ft-lb -. W= 62700 ft-lb (93144.7 N-m) + N-m

$$W = \frac{f_2 \times f}{2} \times \frac{1}{12}$$

$$69700 = \frac{f_2 \times f_2}{2} \times \frac{1}{12}$$

where
$$t \ge 1$$
 in $(ar) \frac{1}{12} ft$

$$(43144.7 = \frac{1}{2} \times \frac{1}{12} = \frac{93144.7 \times 2000}{12000}$$

$$= 7339566.3 \times 1000$$

Feguals the product of the place area sheared and the altimate shearing strength

l = length of cut t = peade trickness

Sus ultimate sheaving strength of material

$$F = l + Su \qquad given$$

$$1650000 = l \times (1 \text{ in}) \times 40,000 \qquad \text{fultimale}$$

$$Shearing}$$

$$l = \frac{1650000}{1 \times 40,000} = 41.75 \text{ in} \qquad \text{Attensith of}$$

$$(ar) 104.3 \text{ cm}$$

In this problem

- Rim of wheel
- Shearing process
- Kinetic energy

N=
$$\frac{Wv^2}{2g}$$

Average force during

Stamping = $\frac{F+o}{2} = \frac{f}{2}$

work done = Product of $\frac{f}{2}$ and thickness' $\frac{f}{2}$

= $\frac{f}{2} \times t$

F = Plade area \times altimate shearing Sheared \times Strength

F = $\frac{f}{2} \times t$

Additional notes

Related Calculations The length of cut computed above can be distributed in any form—square, rectangular, circular, or irregular. This method is suitable for computing the energy stored in a flywheel used for any purpose. Use the general procedure in step 2 for computing the principal dimension in blanking, punching, piercing, trimming, bending, forming, drawing, or coining.

All calculations on the next page

Ti = 3.1416, R = Rotational Speed 8/min
D= Distance of the rim horizontal centreline
from centre of rotation (ft)

= 1318.9Kg

$$K = \frac{\omega v^{2}}{2g} = \frac{4010 \times 33 \cdot 2^{2}}{2 \times 32 \cdot 2} = 68790 \text{ ft-ly}$$

$$(07)$$

$$= 1212.9 \times 10.2$$

$$= 93144.7$$

$$2 \times 9.8$$
N-m

Step 4

Shearing that length of

Trickness of plate

- Shewed Area.

Sheared Area = uldimale Sheared Area = uldimale Shearing Strength.

Average force during $Stamping = \frac{F+0}{2} = \frac{f}{2}$

Because of the at stampemerged from metal work done = Product of $\frac{f}{2}$ and thickness $\frac{f}{2}$ = $\frac{f}{2} \times f$

workdome when all flywheel kimetic Energy used

W=K where K=68700 ft-lb

(or)

-. w=68700 ft-lb (93144-7 N-m)

W= fxt where time 1 im (or) to ft $\frac{1}{2}$ W= $\frac{1}{7}$ × $\frac{1}{12}$ $69700 = \frac{f}{2} \times \frac{1}{12}$ = $f = 68700 \times 2 \times 12$ = 16 50000 Rh $93144-7 = \frac{1}{2} \times \frac{1}{12} = F = 93144-7 \times 2 \times 12$ = 7334566.3 M Feguals the product of the place area sheared and Deultimade Shearing Strength F = plade area x altimate shearing Sheared Strength F= ltsu l= length of cut t = plate trickness Sus cultimate sheaving strength of material

This is my worked example for Section 3-Mechanical Engineering (PDF File Page 307)

Electrical

www.mongroupsydney1.com/1.pdf

Then study Section 4-Electrical Engineering (PDF File Page 885)

For every topic, you need to write the short note on what you understand, formula, summary, outlines and at least 2 problems solution (Please note, each problem is solved in short form, you need to clearly reproduce them by step by step)

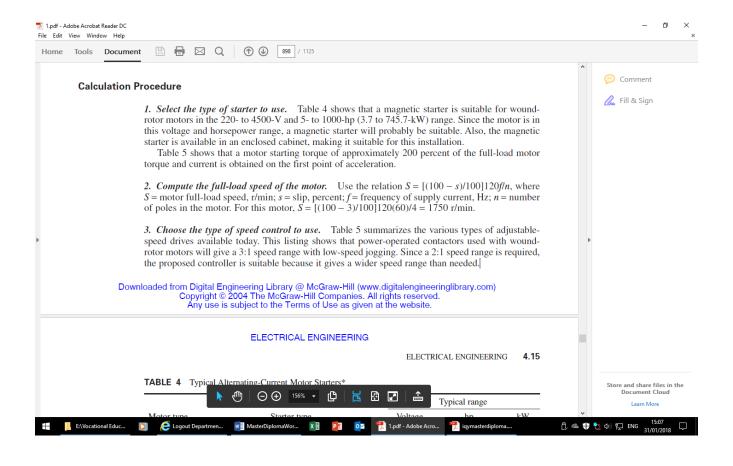
This is my worked example for Selecting Electric-Motor Starting and Speed Controls
Sections,

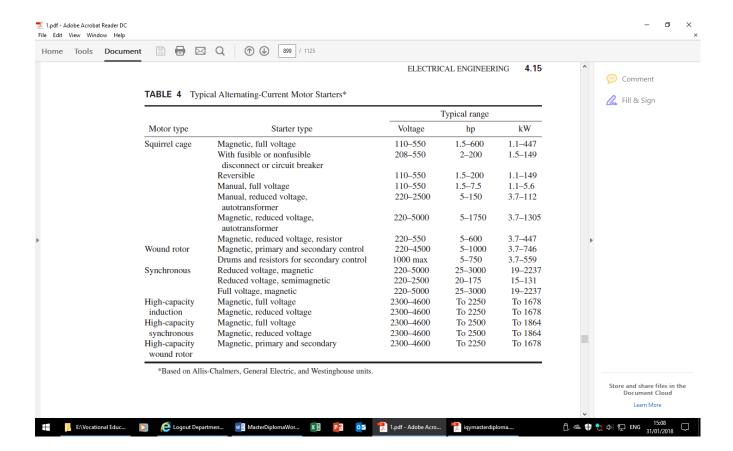
SELECTING ELECTRIC-MOTOR STARTING AND SPEED CONTROLS

Problem

Choose a suitable starter and speed control for a 500-hp (372.8-kW) wound-rotor ac motor that mus thave a speed range of 2 to 1 with a capability for low-speed jogging. The motor is to operate at about 1800 r/min with current supplied at 4160 V, 60 Hz. An enclosed starter and a controller are desirable from the standpoint of protection. What is the actual motor speed if the motor has four poles and a slip of 3 percent?

Outlined Solution





Refer table 4,

wound roter ometer - magnetic, primary and

Lecondary control

Draws and resistors for
Secondary control

for wound notor motors in 220 → 4500v 5 to 1000 HP (3.7 to 745.7 4W)

As this motor is in this voltage and horse power range, a magnetic starter is soutable

motor full load speed

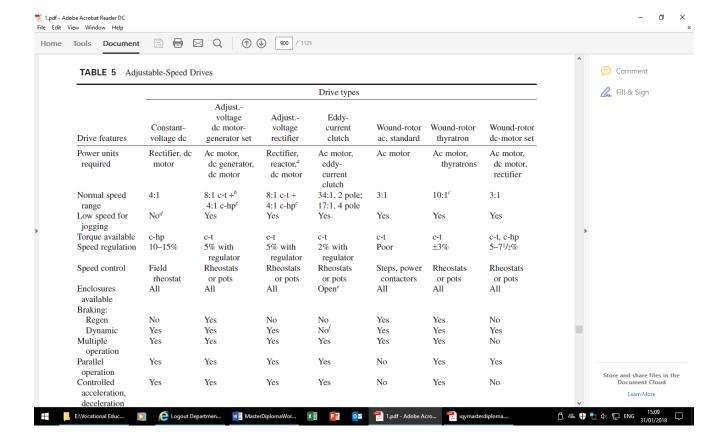
$$S = \left[\frac{(100 - 0)}{100} \right] \times \frac{120f}{9}$$

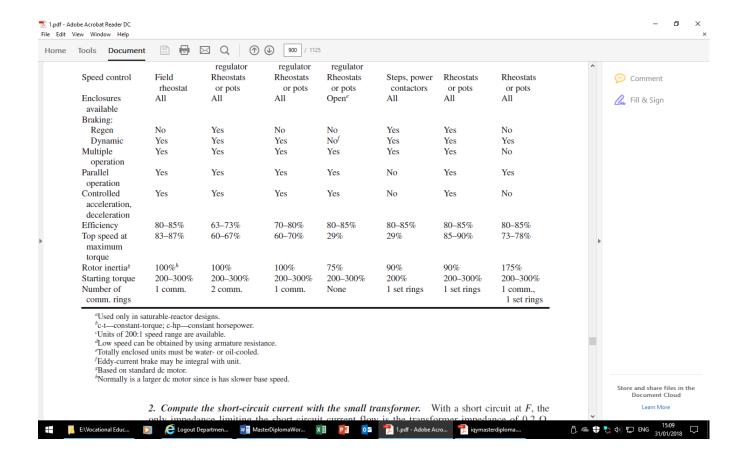
5= motor full load speed 7/min

() = slip (%)

f= frequency of supply current (H2)
m= number of poles in motor

$$S = \frac{(100-3)}{100} \times \frac{120 \times 60}{4} = 1750 \text{ m/m}$$





Type of speed control, Refer Tables

wound roter ac motor

Hormal speed

some 3:1

romge

Low speed for jogging Yes

speed control step/power contactor

as 2:1 speed romge is required, we can choose

power contactor

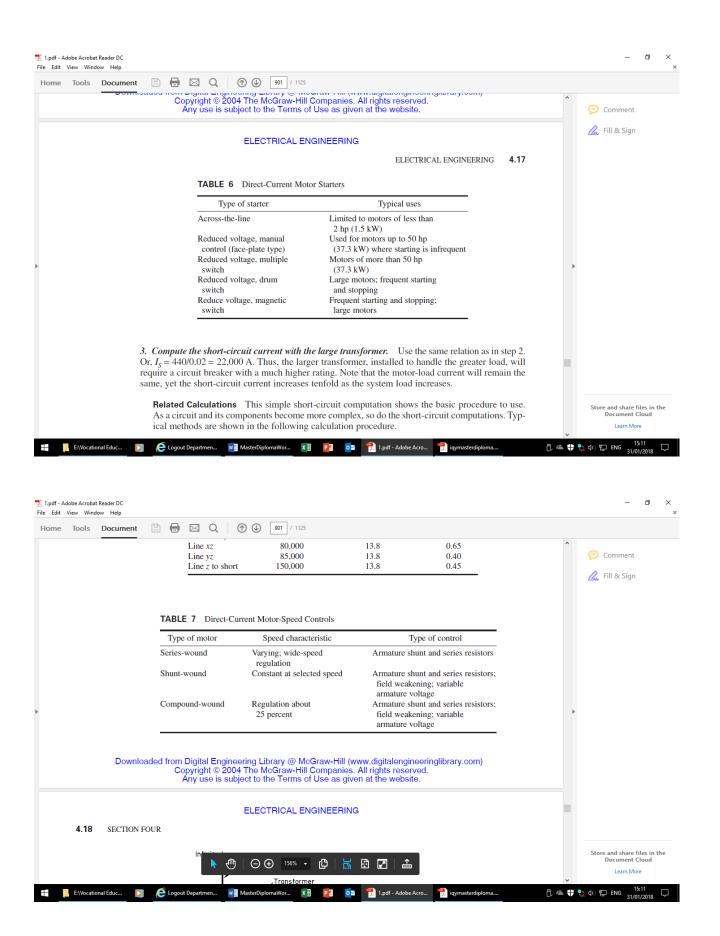
Note from Table 5 that if a wider speed range were required, a thyratron control could produce a range up to 10:1 on a wound-rotor motor. Also, a wound-rotor dc motor set might be used too.

In such an arrangement, an ac and dc motor are combined on the same shaft. The rotor current is converted to dc by external silicon rectifiers and fed back to the dc armature through the commutator.

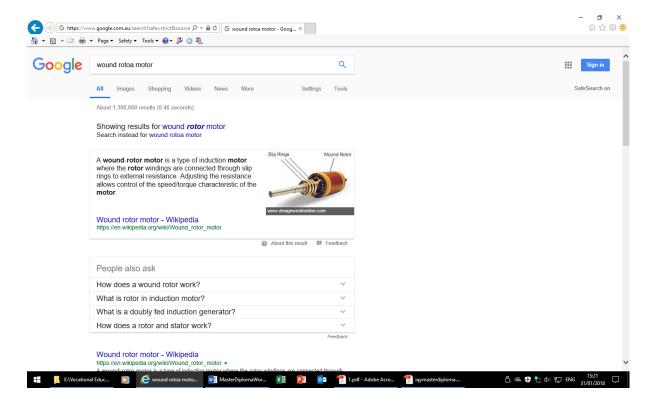
Related Calculations Use the two tables presented here to guide the selection of starters and controls for ac motors serving industrial, commercial, marine, portable, and residential applications.

To choose a dc motor starter, use Table 6 as a guide.

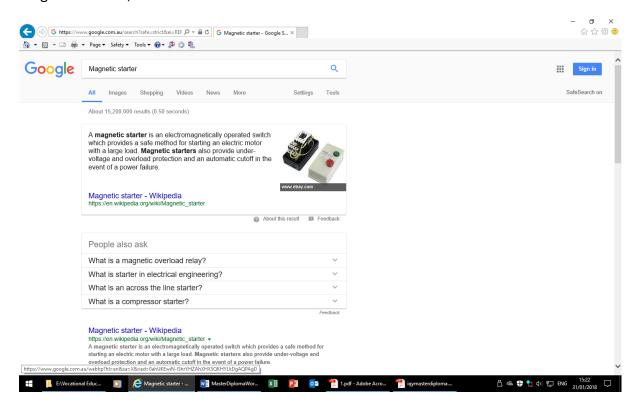
Speed controls for dc motors can be chosen by using Table 7 as a guide. Dc motors are finding increasing use in industry. They are also popular in marine service.



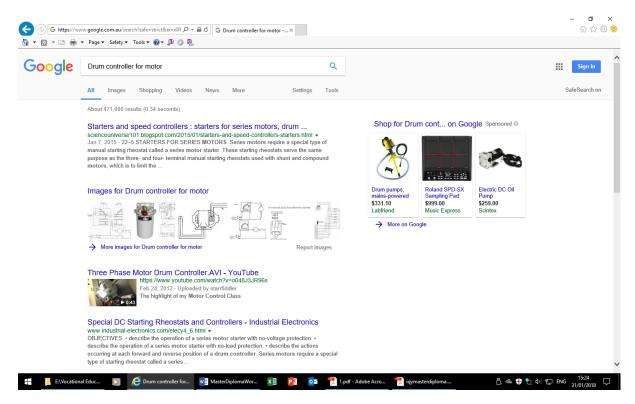
Wound rotor motor? , Do internet search



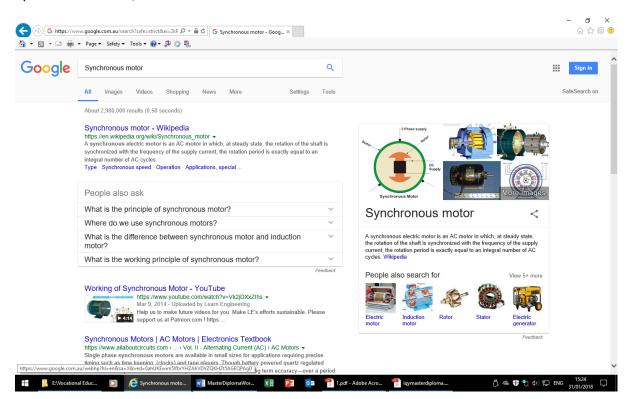
Magnetic starter, Do internet search



Drum control, do Internet search



Synchronous motor, Do internet search



Induction motor, do internet search

