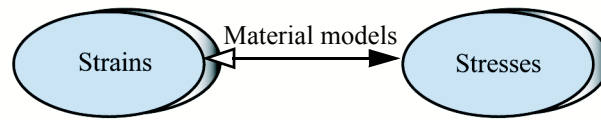


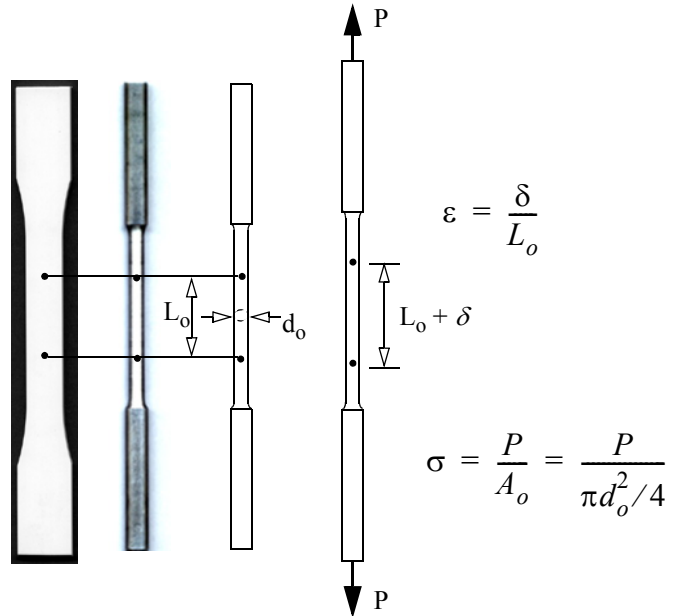
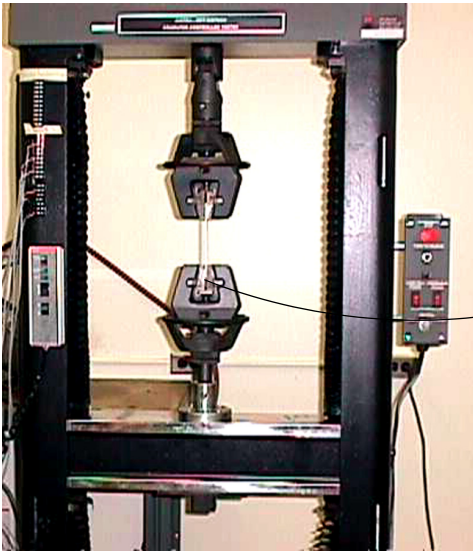
# Mechanical Properties of Materials



## Learning objectives

- Understand the qualitative and quantitative description of mechanical properties of materials.
- Learn the logic of relating deformation to external forces.

# Tension Test



$$\epsilon = \frac{\delta}{L_o} \quad \sigma = \frac{P}{A_o} = \frac{P}{\pi d_o^2 / 4}$$

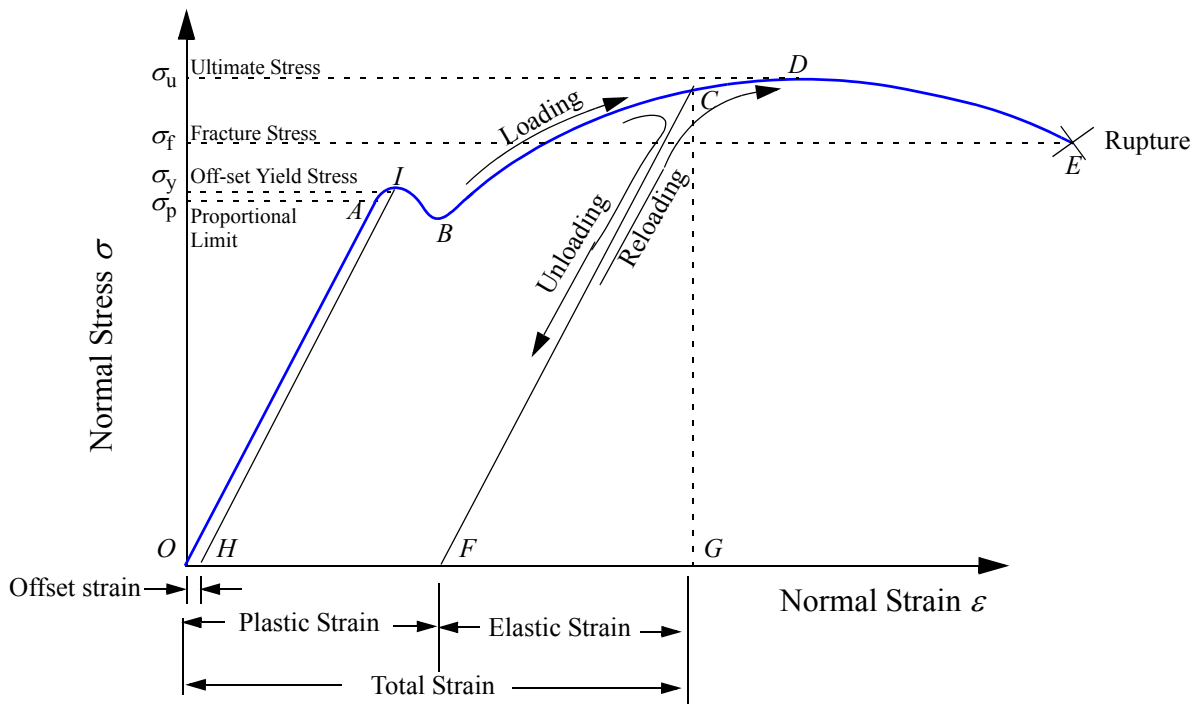


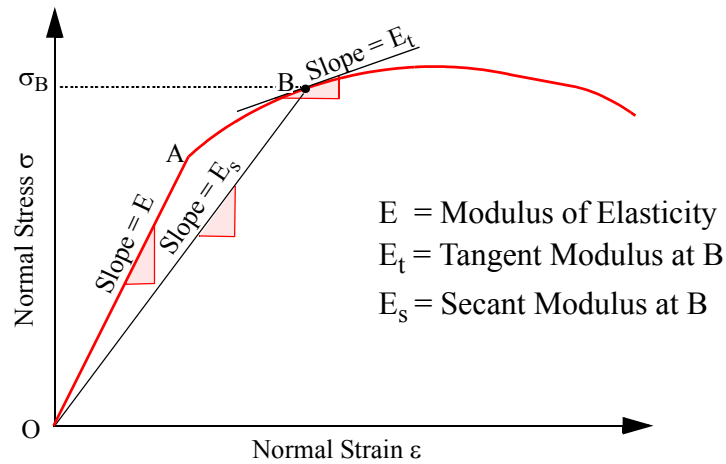
Figure 3.1 Stress-strain curve.

## Definitions

- The point up to which the stress and strain are linearly related is called the **proportional limit**.
- The largest stress in the stress strain curve is called the **ultimate stress**.
- The stress at the point of rupture is called the **fracture or rupture stress**.
- The region of the stress-strain curve in which the material returns to the undeformed state when applied forces are removed is called the **elastic region**.
- The region in which the material deforms permanently is called the **plastic region**.
- The point demarcating the elastic from the plastic region is called the **yield point**. The stress at yield point is called the **yield stress**.
- The permanent strain when stresses are zero is called the **plastic strain**.
- The **off-set yield** stress is a stress that would produce a plastic strain corresponding to the specified off-set strain.
- A material that can undergo large plastic deformation before fracture is called a **ductile material**.
- A material that exhibits little or no plastic deformation at failure is called a **brittle material**.
- **Hardness** is the resistance to indentation.
- The raising of the yield point with increasing strain is called **strain hardening**.
- The sudden decrease in the area of cross-section after ultimate stress is called **necking**.

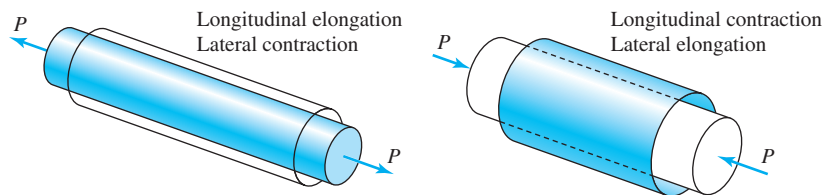


# Material Constants

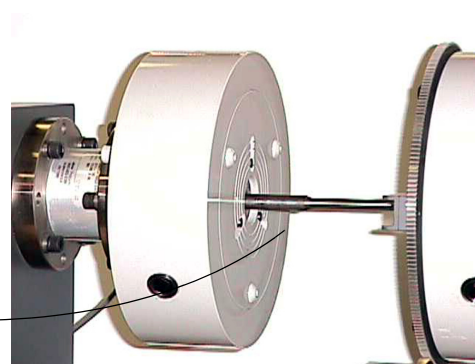
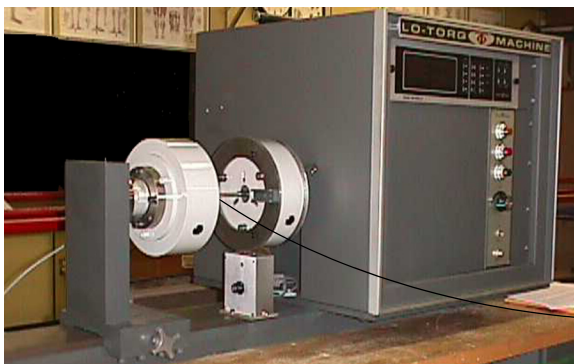


$$\sigma = E\epsilon \text{ ----- Hooke's Law}$$

- $E$  **Young's Modulus or Modulus of Elasticity**



- Poisson's ratio: 
$$\nu = -\left(\frac{\epsilon_{lateral}}{\epsilon_{longitudinal}}\right)$$

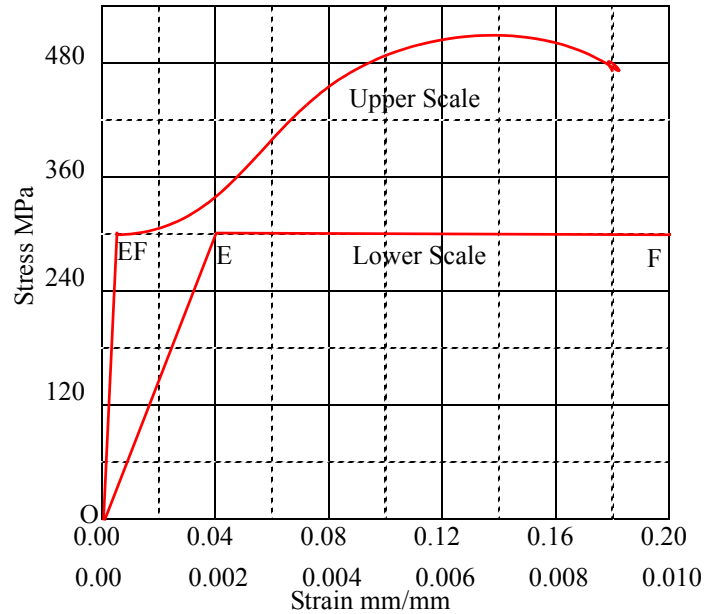
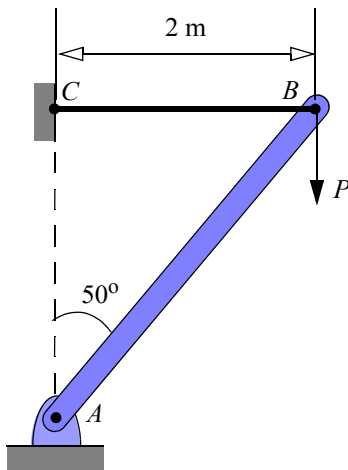


$$\tau = G\gamma$$

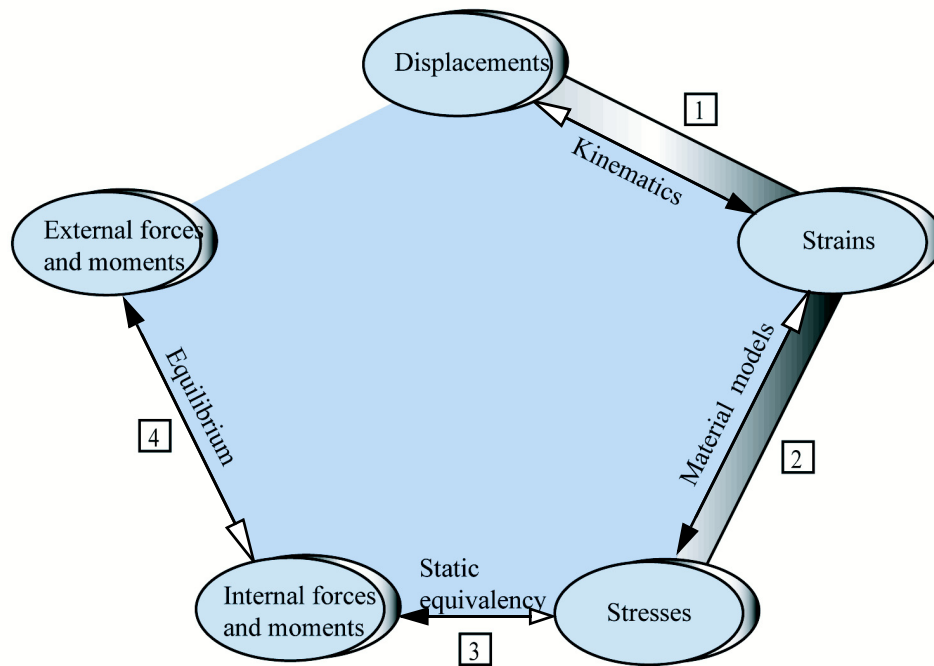
$G$  is called the **Shear Modulus of Elasticity** or the **Modulus of Rigidity**

**C3.1** An aluminum rectangular bar has a cross-section of 25 mm x 50 mm and a length of 500 mm. The Modulus of Elasticity of  $E = 70$  GPa and a Poisson's ratio of  $\nu = 0.25$ . Determine the percentage change in the volume of the bar when an axial force of 300 kN is applied to the bar.

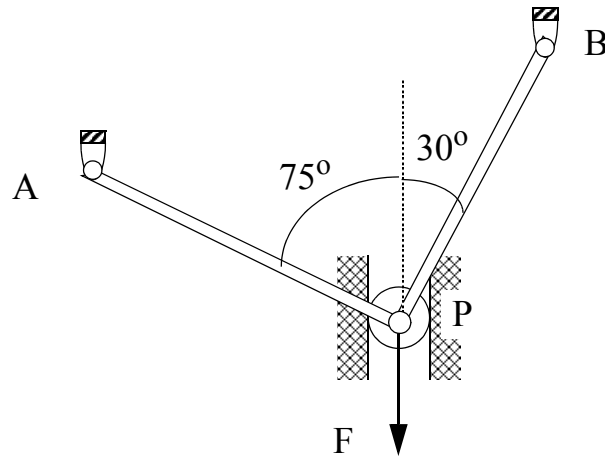
**C3.2** A rigid bar  $AB$  of negligible weight is supported by cable of diameter 5 mm, as shown. The cable is made from a material that has a stress-strain curve shown. (a) Determine the extension of the cable when  $P = 10$  kN. (b) What is the permanent deformation in  $BC$  when the load  $P$  is removed?



# Logic in structural analysis



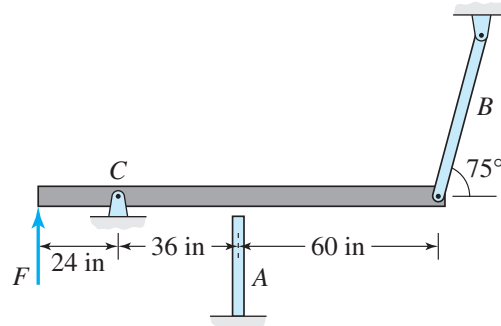
**C3.3** A roller slides in a slot by the amount  $\delta_P = 0.25$  mm in the direction of the force  $F$ . Both bars have an area of cross-section of  $A = 100 \text{ mm}^2$  and a Modulus of Elasticity  $E = 200 \text{ GPa}$ . Bar AP and BP have lengths of  $L_{AP} = 200$  mm and  $L_{BP} = 250$  mm respectively. Determine the applied force  $F$ .



**Fig. C3.3**



**C3.4** A gap of 0.004 in. exists between a rigid bar and bar  $A$  before a force  $F$  is applied (Figure P3.4). The rigid bar is hinged at point  $C$ . Due to force  $F$  the strain in bar  $A$  was found to be  $-500 \mu\text{in/in}$ . The lengths of bars  $A$  and  $B$  are 30 in. and 50 in., respectively. Both bars have cross-sectional areas  $A = 1 \text{ in.}^2$  and a modulus of elasticity  $E = 30,000 \text{ ksi}$ . Determine the applied force  $F$ .

**Fig. C3.4****Fig. C3.4**

## Isotropy and Homogeneity

Linear relationship between stress and strain components:

$$\varepsilon_{xx} = C_{11}\sigma_{xx} + C_{12}\sigma_{yy} + C_{13}\sigma_{zz} + C_{14}\tau_{yz} + C_{15}\tau_{zx} + C_{16}\tau_{xy}$$

$$\varepsilon_{yy} = C_{21}\sigma_{xx} + C_{22}\sigma_{yy} + C_{23}\sigma_{zz} + C_{24}\tau_{yz} + C_{25}\tau_{zx} + C_{26}\tau_{xy}$$

$$\varepsilon_{zz} = C_{31}\sigma_{xx} + C_{32}\sigma_{yy} + C_{33}\sigma_{zz} + C_{34}\tau_{yz} + C_{35}\tau_{zx} + C_{36}\tau_{xy}$$

$$\gamma_{yz} = C_{41}\sigma_{xx} + C_{42}\sigma_{yy} + C_{43}\sigma_{zz} + C_{44}\tau_{yz} + C_{45}\tau_{zx} + C_{46}\tau_{xy}$$

$$\gamma_{zx} = C_{51}\sigma_{xx} + C_{52}\sigma_{yy} + C_{53}\sigma_{zz} + C_{54}\tau_{yz} + C_{55}\tau_{zx} + C_{56}\tau_{xy}$$

$$\gamma_{xy} = C_{61}\sigma_{xx} + C_{62}\sigma_{yy} + C_{63}\sigma_{zz} + C_{64}\tau_{yz} + C_{65}\tau_{zx} + C_{66}\tau_{xy}$$

- An **isotropic material** has a stress-strain relationships that are independent of the orientation of the coordinate system at a point.
- A material is said to be **homogenous** if the material properties are the same at all points in the body. Alternatively, if the material constants  $C_{ij}$  are functions of the coordinates  $x$ ,  $y$ , or  $z$ , then the material is called non-homogenous.

For Isotropic Materials: 
$$G = \frac{E}{2(1 + \nu)}$$

# Generalized Hooke's Law for Isotropic Materials

- The relationship between stresses and strains in three-dimensions is called the **Generalized Hooke's Law**.

$$\varepsilon_{xx} = [\sigma_{xx} - \nu(\sigma_{yy} + \sigma_{zz})]/E$$

$$\varepsilon_{yy} = [\sigma_{yy} - \nu(\sigma_{zz} + \sigma_{xx})]/E$$

$$\varepsilon_{zz} = [\sigma_{zz} - \nu(\sigma_{xx} + \sigma_{yy})]/E$$

$$\gamma_{xy} = \tau_{xy}/G$$

$$\gamma_{yz} = \tau_{yz}/G$$

$$\gamma_{zx} = \tau_{zx}/G$$

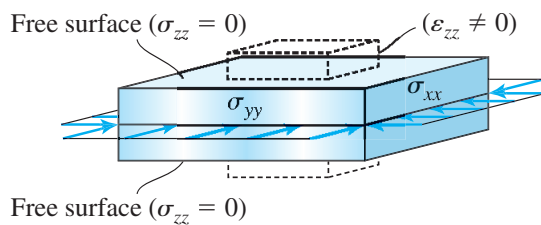
$$G = \frac{E}{2(1 + \nu)}$$

$$\begin{Bmatrix} \varepsilon_{xx} \\ \varepsilon_{yy} \\ \varepsilon_{zz} \end{Bmatrix} = \frac{1}{E} \begin{bmatrix} 1 & -\nu & -\nu \\ -\nu & 1 & -\nu \\ -\nu & -\nu & 1 \end{bmatrix} \begin{Bmatrix} \sigma_{xx} \\ \sigma_{yy} \\ \sigma_{zz} \end{Bmatrix}$$

# Plane Stress and Plane Strain

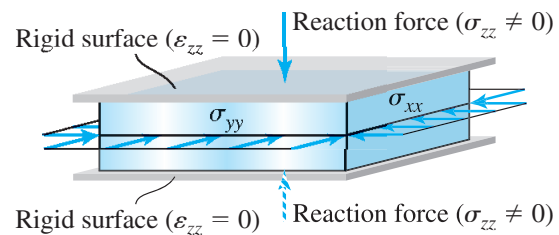
$$\begin{array}{lcl}
 \text{Plane Stress} \longrightarrow & \begin{bmatrix} \sigma_{xx} & \tau_{xy} & 0 \\ \tau_{yx} & \sigma_{yy} & 0 \\ 0 & 0 & 0 \end{bmatrix} & \xrightarrow{\text{Generalized Hooke's Law}} \begin{bmatrix} \epsilon_{xx} & \gamma_{xy} & 0 \\ \gamma_{yx} & \epsilon_{yy} & 0 \\ 0 & 0 & \epsilon_{zz} = -\frac{\nu}{E}(\sigma_{xx} + \sigma_{yy}) \end{bmatrix} \\
 \text{Plane Strain} \longrightarrow & \begin{bmatrix} \epsilon_{xx} & \gamma_{xy} & 0 \\ \gamma_{yx} & \epsilon_{yy} & 0 \\ 0 & 0 & 0 \end{bmatrix} & \xrightarrow{\text{Generalized Hooke's Law}} \begin{bmatrix} \sigma_{xx} & \tau_{xy} & 0 \\ \tau_{yx} & \sigma_{yy} & 0 \\ 0 & 0 & \sigma_{zz} = \nu(\sigma_{xx} + \sigma_{yy}) \end{bmatrix}
 \end{array}$$

## Plane Stress



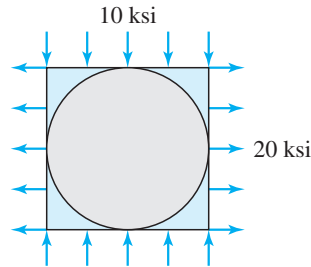
(a)

## Plane Strain



(b)

**C3.5** A 2 in x 2 in square with a circle inscribed is stressed as shown Fig. C3.5. The plate material has a Modulus of Elasticity of  $E = 10,000$  ksi and a Poisson's ratio  $\nu = 0.25$ . Determine the major and minor axis of the ellipse formed due to deformation assuming (a) plane stress. (b) plane strain.



**Fig. C3.5**

## Class Problem 1

The stress components at a point are as given.

Determine  $\epsilon_{xx}$  assuming (a) Plane stress (b) Plane strain

$$\sigma_{xx} = 100 \text{ MPa}(T)$$

$$\sigma_{yy} = 200 \text{ MPa}(C)$$

$$\tau_{xy} = -125 \text{ MPa}$$

$$E = 200 \text{ GPa}$$

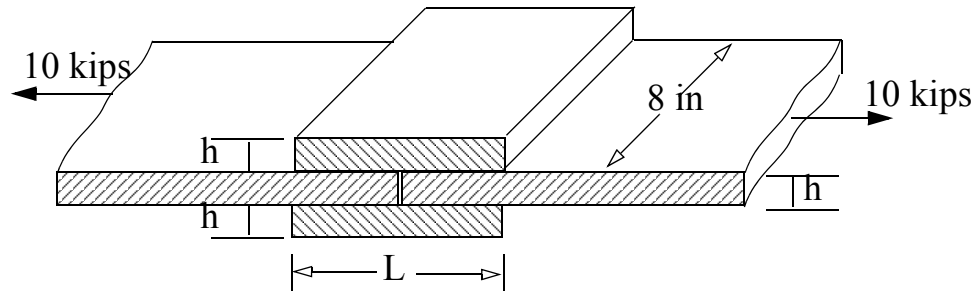
$$\nu = 0.25$$

## Failure and factor of safety

- Failure implies that a component or a structure does not perform the function it was designed for.

$$K_{safety} = \frac{\text{Failure producing value}}{\text{Computed(allowable)value}} \quad 3.1$$

**C3.6** An adhesively bonded joint in wood is fabricated as shown. For a factor of safety of 1.25, determine the minimum overlap length  $L$  and dimension  $h$  to the nearest  $1/8$ th inch. The shear strength of adhesive is 400 psi and the wood strength is 6 ksi in tension.



**Fig. C3.6**

## Common Limitations to Theories in Chapters 4-7

- The length of the member is significantly greater (approximately 10 times) than the greatest dimension in the cross-section.
- We are away from regions of stress concentration, where displacements and stresses can be three-dimensional.
- The variation of external loads or changes in the cross-sectional area is gradual except in regions of stress concentration.
- The external loads are such that the axial, torsion and bending problems can be studied individually.